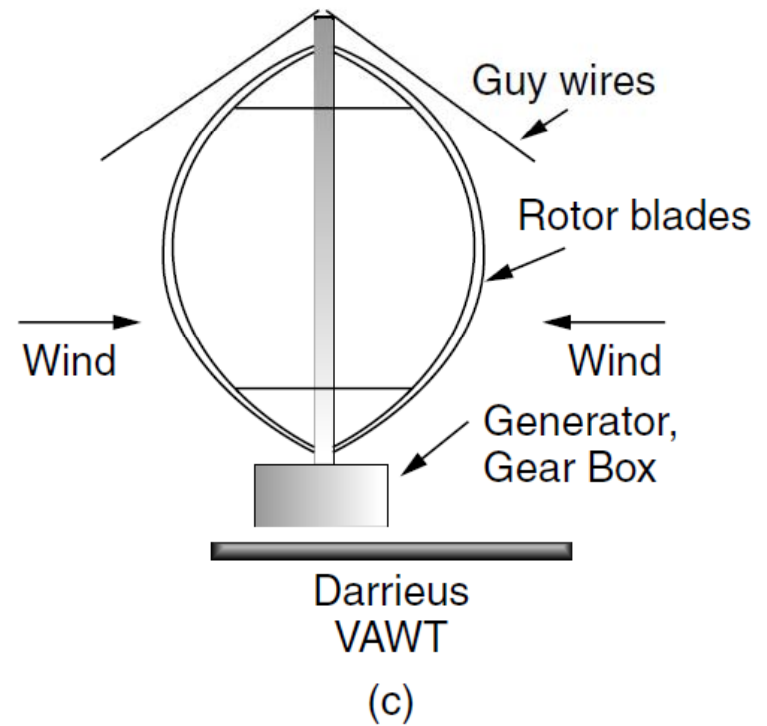
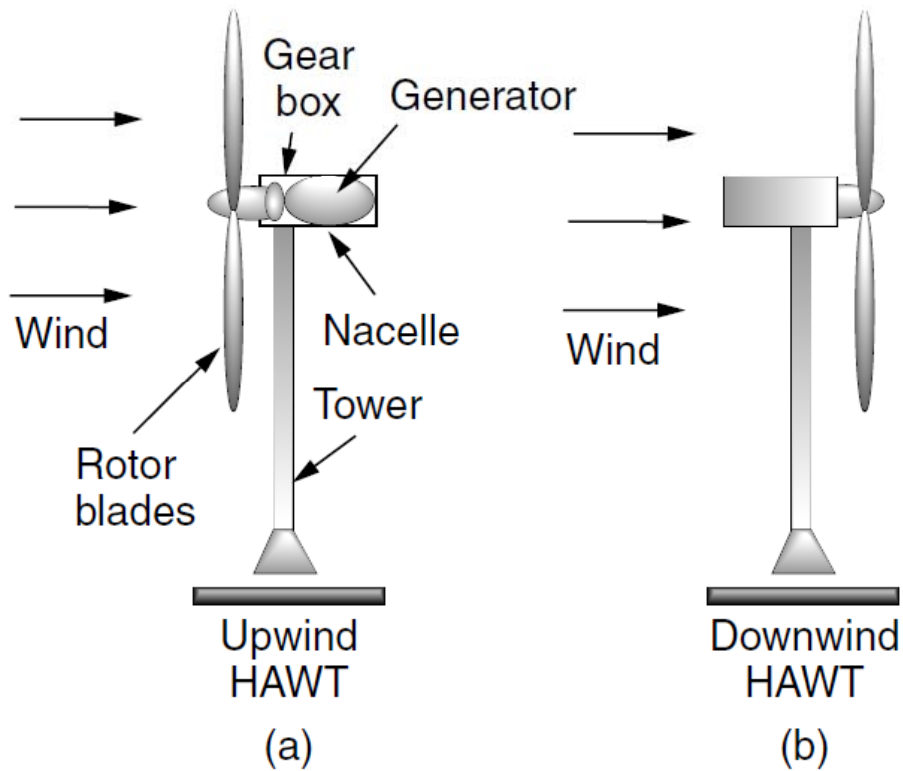


# Renewable Energy

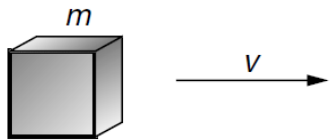


# WIND POWER SYSTEMS



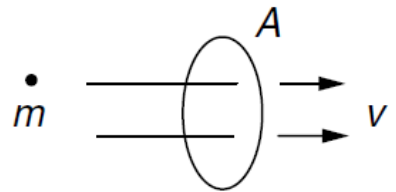
# POWER IN THE WIND





A 3D cube is shown with the letter  $m$  above it. To its right is a horizontal arrow pointing to the right, labeled with the letter  $v$ .

$$\text{K.E.} = \frac{1}{2}mv^2$$



A dot representing mass  $\dot{m}$  is on the left. Two horizontal arrows point to the right, passing through an oval representing an area  $A$ . To the right of the oval is another horizontal arrow labeled  $v$ .

$$\text{Power through area } A = \frac{\text{Energy}}{\text{Time}} = \frac{1}{2} \left( \frac{\text{Mass}}{\text{Time}} \right) v^2$$

$$\left( \frac{\text{Mass passing through } A}{\text{Time}} \right) = \dot{m} = \rho Av$$

$$P_w = \frac{1}{2} \rho A v^3$$

*In S.I. units;  $P_w$  is the power in the wind (watts);  $\rho$  is the air density ( $\text{kg}/\text{m}^3$ ) (at  $15^\circ\text{C}$  and  $1 \text{ atm}$ ,  $\rho = 1.225 \text{ kg}/\text{m}^3$ );  $A$  is the cross-sectional area through which the wind passes ( $\text{m}^2$ ); and  $v =$  windspeed normal to  $A$  ( $\text{m}/\text{s}$ ) (a useful conversion:  $1 \text{ m}/\text{s} = 2.237 \text{ mph}$ ).*

# Temperature Correction for Air Density



ideal gas law  $PV = nRT$

where  $P$  is the absolute pressure (atm),  $V$  is the volume ( $m^3$ ),  $n$  is the mass (mol),  $R$  is the ideal gas constant =  $8.2056 \times 10^{-5} m^3 \cdot atm \cdot K^{-1} \cdot mol^{-1}$ , and  $T$  is the absolute temperature (K), where  $K = ^\circ C + 273.15$ .

$$\rho(\text{kg}/\text{m}^3) = \frac{n(\text{mol}) \cdot \text{M.W.}(\text{g}/\text{mol}) \cdot 10^{-3}(\text{kg}/\text{g})}{V(\text{m}^3)}$$

$$\rho = \frac{P \times \text{M.W.} \times 10^{-3}}{RT}$$

Find the density of air at 1 atm and 30°C (86°F)

$$\rho = \frac{1 \text{ atm} \times 28.97 \text{ g/mol} \times 10^{-3} \text{ kg/g}}{8.2056 \times 10^{-5} \text{ m}^3 \cdot \text{atm}/(\text{K} \cdot \text{mol}) \times (273.15 + 30) \text{ K}} = 1.165 \text{ kg}/\text{m}^3$$

M.W. stand for the molecular weight of the gas (g/mol), Air=28.97 g/mol

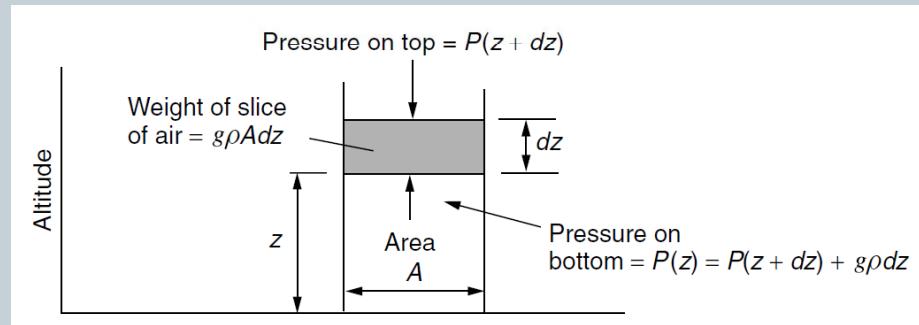
# Density of Dry Air at a Pressure of 1 Atmosphere



Temperature (°C)	Temperature (°F)	Density (kg/m <sup>3</sup> )	Density Ratio ( $K_T$ )
-15	5.0	1.368	1.12
-10	14.0	1.342	1.10
-5	23.0	1.317	1.07
0	32.0	1.293	1.05
5	41.0	1.269	1.04
10	50.0	1.247	1.02
<b>15</b>	<b>59.0</b>	<b>1.225</b>	1.00
20	68.0	1.204	0.98
25	77.0	1.184	0.97
30	86.0	1.165	0.95
35	95.0	1.146	0.94
40	104.0	1.127	0.92

<sup>a</sup>The density ratio  $K_T$  is the ratio of density at  $T$  to the density at the standard (boldfaced) 15°C.

# Altitude Correction for Air Density



$$dP = P(z + dz) - P(z) = -g \rho dz$$

$$\frac{dP}{dz} = -\rho g$$

$$\rho = \frac{P \times \text{M.W.} \times 10^{-3}}{RT}$$

$$\frac{dP}{dz} = - \left( \frac{g \text{ M.W.} \times 10^{-3}}{R \cdot T} \right) \cdot P$$

$$\frac{dP}{dz} = - \left[ \frac{9.806(\text{m/s}^2) \times 28.97(\text{g/mol}) \times 10^{-3}(\text{kg/g})}{8.2056 \times 10^{-5}(\text{m}^3 \cdot \text{atm} \cdot \text{K}^{-1} \cdot \text{mol}^{-1}) \times 288.15 \text{ K}} \right]$$

$$\times \left( \frac{\text{atm}}{101,325 \text{ Pa}} \right) \cdot \left( \frac{1 \text{ Pa}}{\text{N/m}^2} \right) \left( \frac{1 \text{ N}}{\text{kg} \cdot \text{m/s}^2} \right) \cdot P$$

$$\frac{dP}{dz} = -1.185 \times 10^{-4} P$$

$$P = P_0 e^{-1.185 \times 10^{-4} H} = 1(\text{atm}) \cdot e^{-1.185 \times 10^{-4} H}$$



- Find the air density (a), at 15°C (288.15 K), at an elevation of 2000 m (6562 ft). Then (b) find it assuming an air temperature of 5°C at 2000 m.

$$P = 1 \text{ atm} \times e^{-1.185 \times 10^{-4} \times 2000} = 0.789 \text{ atm}$$

$$\begin{aligned}\rho &= \frac{P \cdot \text{M.W.} \cdot 10^{-3}}{R \cdot T} \\ &= \frac{0.789(\text{atm}) \times 28.97(\text{g/mol}) \times 10^{-3}(\text{kg/g})}{8.2056 \times 10^{-5}(\text{m}^3 \cdot \text{atm} \cdot \text{K}^{-1} \cdot \text{mol}^{-1}) \times 288.15 \text{ K}} \\ &= 0.967 \text{ kg/m}^3\end{aligned}$$

At 5°C and 2000 m, the air density would be

$$\begin{aligned}\rho &= \frac{0.789(\text{atm}) \times 28.97(\text{g/mol}) \times 10^{-3}(\text{kg/g})}{8.2056 \times 10^{-5}(\text{m}^3 \cdot \text{atm} \cdot \text{K}^{-1} \cdot \text{mol}^{-1}) \times (273.15 + 5) \text{ K}} \\ &= 1.00 \text{ kg/m}^3\end{aligned}$$



$$\rho = 1.225 K_T K_A$$

Altitude (meters)	Altitude (feet)	Pressure (atm)	Pressure Ratio ( $K_A$ )
0	0	1	1
200	656	0.977	0.977
400	1312	0.954	0.954
600	1968	0.931	0.931
800	2625	0.910	0.910
1000	3281	0.888	0.888
1200	3937	0.868	0.868
1400	4593	0.847	0.847
1600	5249	0.827	0.827
1800	5905	0.808	0.808
2000	6562	0.789	0.789
2200	7218	0.771	0.771

Temperature (°C)	Temperature (°F)	Density (kg/m <sup>3</sup> )	Density Ratio ( $K_T$ )
-15	5.0	1.368	1.12
-10	14.0	1.342	1.10
-5	23.0	1.317	1.07
0	32.0	1.293	1.05
5	41.0	1.269	1.04
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# IMPACT OF TOWER HEIGHT



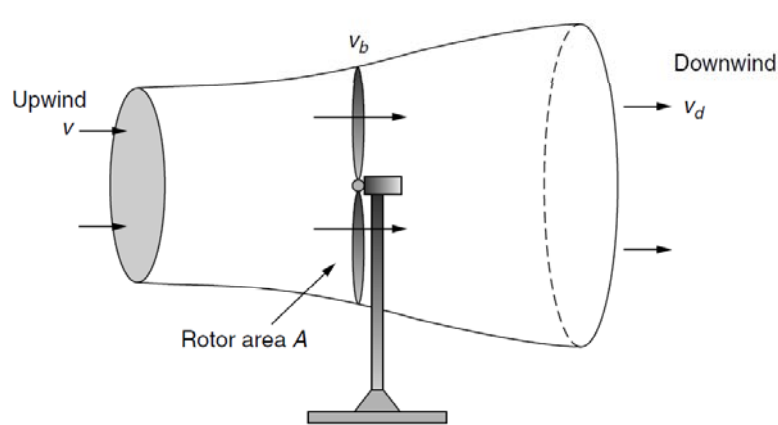
$$\left(\frac{v}{v_0}\right) = \left(\frac{H}{H_0}\right)^\alpha$$

$$\left(\frac{v}{v_0}\right) = \frac{\ln(H/z)}{\ln(H_0/z)}$$

Terrain Characteristics	Friction Coefficient $\alpha$
Smooth hard ground, calm water	0.10
Tall grass on level ground	0.15
High crops, hedges and shrubs	0.20
Wooded countryside, many trees	0.25
Small town with trees and shrubs	0.30
Large city with tall buildings	0.40

Roughness Class	Description	Roughness Length $z(m)$
0	Water surface	0.0002
1	Open areas with a few windbreaks	0.03
2	Farm land with some windbreaks more than 1 km apart	0.1
3	Urban districts and farm land with many windbreaks	0.4
4	Dense urban or forest	1.6

# MAXIMUM ROTOR EFFICIENCY



$$P_b = \frac{1}{2} \dot{m} (v^2 - v_d^2)$$

$$\dot{m} = \rho A v_b$$

$$P_b = \frac{1}{2} \rho A \left( \frac{v + v_d}{2} \right) (v^2 - v_d^2) \quad \lambda = \left( \frac{v_d}{v} \right)$$

$$P_b = \frac{1}{2} \rho A \left( \frac{v + \lambda v}{2} \right) (v^2 - \lambda^2 v^2) = \underbrace{\frac{1}{2} \rho A v^3}_{\text{Power in the wind}} \cdot \underbrace{\left[ \frac{1}{2} (1 + \lambda)(1 - \lambda^2) \right]}_{\text{Fraction extracted}}$$

$$P_b = \frac{1}{2} \rho A v^3 \cdot C_p$$

$$\text{Rotor efficiency} = C_p = \frac{1}{2} (1 + \lambda)(1 - \lambda^2)$$

$$\begin{aligned} \frac{dC_p}{d\lambda} &= \frac{1}{2} [(1 + \lambda)(-2\lambda) + (1 - \lambda^2)] = 0 \\ &= \frac{1}{2} [(1 + \lambda)(-2\lambda) + (1 + \lambda)(1 - \lambda)] = \frac{1}{2} (1 + \lambda)(1 - 3\lambda) = 0 \end{aligned}$$

$$\lambda = \frac{v_d}{v} = \frac{1}{3}$$

$$\text{Maximum rotor efficiency} = \frac{1}{2} \left( 1 + \frac{1}{3} \right) \left( 1 - \frac{1}{3^2} \right) = \frac{16}{27} = 0.593 = 59.3\%$$

# tip-speed ratio



$$\text{Tip-Speed-Ratio (TSR)} = \frac{\text{Rotor tip speed}}{\text{Wind speed}} = \frac{\text{rpm} \times \pi D}{60 v}$$

where rpm is the rotor speed, revolutions per minute;  $D$  is the rotor diameter (m); and  $v$  is the wind speed (m/s) upwind of the turbine.

